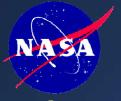


Chief Engineer's Council
Montreal
8/23-24/2016

Ken Welzyn - NASA MSFC Jeff Pilet - Lockheed Martin



External Tank Legacy of Success



• Agenda

- External Tank Overview
- Super Lightweight Tank Verification
- Return to Flight
- Engine Cut-Off Sensor Circuit
- ET-124 Hail Damage Recovery
- STS-130 / ET-134 Launch



STS-132/ ET-136 05/14/10



External Tank Legacy of Success Overview



Notable Events

- ATP 1972
- 1st Production Article 1978
- 1st Flight Article Complete 1979
- 1st SLWT Complete Enabled access ISS 1998
- TPS design changes implemented post-Columbia (RTF) to reduce debris 2003
- All manifested tanks completed and delivered to KSC 2010
- 139 tanks manufactured in total

1980's

1990's

2000's

1981 - 1983

Tank structure 'evolved' to improve payload performance and producibility

HWT6 flown

1981 - 1998

86 flown

1998 - Present

2003 - 2009

Changes

Implemented to **Reduce Debris**

External Tank Evolution / Weight

- HWT (1981 1983): 77,086 lb. (6 flown)
- LWT (1981 1998): 66,000 lb. (86 flown)
- SLWT (1998 present): 58,550 lb. (40 flown)

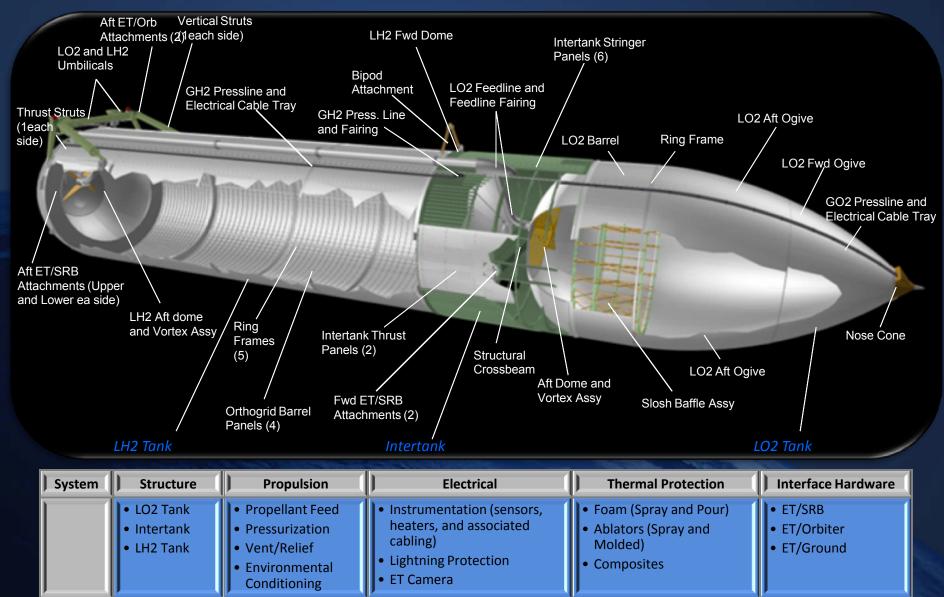
21 flown

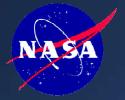
19 flown



External Tank Legacy of Success Overview







External Tank Legacy of Success Overview



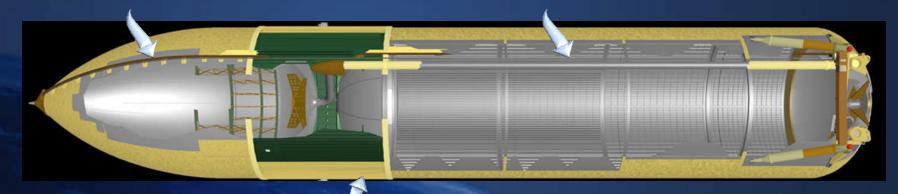
• Interesting ET Facts

Liquid Oxygen Tank

- 1,385,000 lbs. / 145,138 gallons Oxidizer
- -297°F

Liquid Hydrogen Tank

- 231,000 lbs. / 309,139 gallons fuel
- -423° F



Intertank

- Unpressurized Structure
- ~4,000 lbs. of thermal protection materials (16,750 sq. ft.)
- ~38 Miles of Electrical Wiring
- *Length* = 153.8 Feet
- *Diameter* = 27.6
- > 1/2 mile of pressure vessel welds

- Max TPS thickness ~2.5" (LO2)
- Min TPS thickness ~0.2" (Intertank)
- Max Al substrate thickness ~2.0" (Intertank)
- Min Al substrate thickness ~0.050" (Intertank)
- Max LO2 operating pressure ~70 psig
- Max LH2 operating pressure ~40 psig





Goal

- Optimize External Tank structural mass to support ISS construction
 - ~7500 lbm required to achieve 51.6° orbital inclination with ISS payload
 - Super Lightweight Tank (SLWT) program initiated to provide performance

Challenges

- Required parallel development of lightweight aluminum-lithium material, and associated manufacturing processes, and design
- Aggressive schedule to support ISS program
- Structural verification program constrained by funding and schedule
 - Dedicated full-scale, cryogenic STAs not planned
- Significant production impacts caused by Al-Li alloy weld-related rework

How'd We Do It?

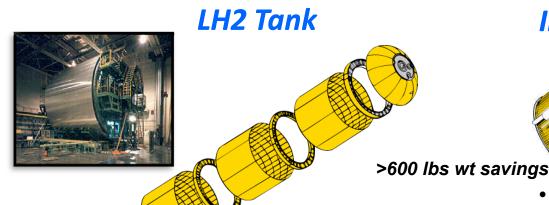
- Leveraged government and corporate research with Al-Li alloys
- Used new orthogrid design for LH2 tank barrels to optimize performance
- Developed innovative design / material verification and acceptance test program
- Fully engaged industry experts and technical community early in design verification
- Evolved design to mitigate production issues

NASA

External Tank Legacy of Success Super Lightweight Tank

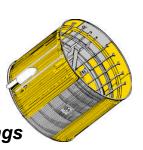


• Super Lightweight Tank (SLWT) Change from Lightweight (LWT)



>4000 lbs wt savings

Intertank





• Substitute Al 2090 for Al 2024 and Al 7075

• Substitute Al 2195 for Al 2219

• Redesign to Orthogrid Waffle

• Optimize TPS Application

= Al Li 2090, 2195

= Other Redesigned Parts







• Substitute Al 2195 for Al 2219

Resize





- Major configuration change implemented on SLWT LH2 Tank barrels
 - Was: Al 2219 T-stiffened
 - Now: Al 2195 Orthogrid
 - Required development of new manufacturing process for machining, forming, and welding



Standard Weight (SWT) and Lightweight (LWT) - Al 2219 alloy



Super Lightweight (SLWT) – Al 2195 alloy





- Innovative structural verification plan established for SLWT
 - An independent Verification Team was formed with industry experts
 - Verification Team established plan that verified each failure mode by either test, flight history, or independent analyses
 - Team utilized wealth of data from SWT and LWT heritage
 - STA, GVTA, MPTA, DDT&E and 90 flights

ET Structural Verification Approach

Design Verification
Partial / Complimentary



"Test what you fly - Fly what you test"

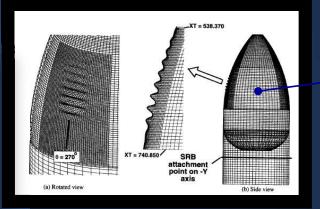


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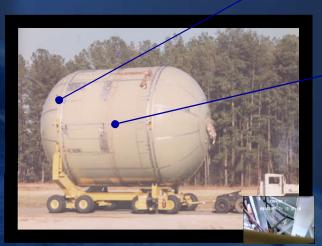
SLWT Verification Process

	Existing Data Base	Component Tests	Independent Stability Analysis	ALTA	Proof Tests	Protoflight Tests	Tanking Test	Engineering Analysis
LO2 Tank Structure								
Stability	х	Х	x	Х				х
Strength	х				х			х
Intertank Structure								
Stability	х	х	x					х
Strength	х							х
LH2 tank Structure								
Stability	х	х		х		х	х	х
Strength	х	х			х	х		х
Thermal Protection Systems	x	х					х	х
MPS / Electrical Systems		х					х	х
Interface / Component Hdwr	х							х

ALTA used to verify multiple hardware elements and failure modes
- - LH2 Barrel Panels, LO2 Dome, Fusion Welding



LO2 Tank Independent Stability Analysis (LaRC)



ALTA at MSFC

Test-based Verification Performed for All Hardware and Failure Modes
- - Program Mitigated Requirement for Full-scale Cryogenic Test





- ALTA Capability Test
 - Structure tested to demonstrated to > ultimate load











- SLWT design further evolved to mitigate production issues with fusion welding and maintain payload performance
 - Reverted back to Al 2219 for domes / ogives with structural optimization (+wt)
 - Further optimized LH2 tank in areas with high margins (-wt)
 - Implemented Al-Li in non-welded application (Intertank thrust panel) (-wt)

ET-123 / STS-116 (first flown)

- Substituted Al 2195 LO₂ fwd ogive, LH₂ fwd & aft dome gores with Al 2219 / resize membrane
- Redesigned Intertank thrust panels

Legend

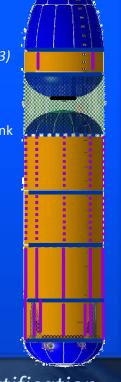
- Al 2195
- AI 2219
- Design changed (thickness optimized / material unchanged)
- Al 2297
- FSW on ET-132 & subs
 - FSW on ET-134 & subs

ET-127 / STS-119 (first flown)

- Substituted Al 2195 LO₂ dome gores with Al 2219 / resize membrane
- Resized LO₂ tank X_T 851 frame webs, LO₂ dome car and LH2 barrels 1 and 2

ET-134 / STS-130 (first flown)

- Implemented FSW on LO₂ & LH₂ Tank barrel longitudinal welds (barrels 3 & 4 only for ETs 132 & 133)
- Substituted Al 2195 LO₂ aft ogive with Al 2219 / resize membrane
- thrust panels with Al 2297





- Substituted Al 2219 Intertank





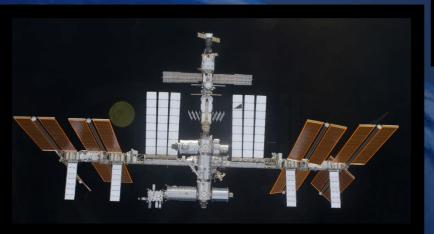
4

Results

- 1st SLWT tank delivered in 44 months
- SLWT tanks delivered on-time and within budget and have performed flawlessly
- SLWT tank enabled ISS construction by improving Shuttle payload capability
- ISS construction required international collaboration
- SLWT tank + SSP + ISS = World Peace!!



STS-91 / ET-96



International Space Station photographed by STS-132 crew



International Teamwork made possible by SLWT



Improved Relations between
Nuclear Super Powers

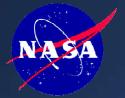






- Engage industry experts early in design verification cycle
- Verification program should be test-based and failure mode specific
- Tests to design capability are critical to understand margins
- Tests should be performed incrementally to reduce program risk
 - Component Large scale Acceptance
- Protoflight tests can be used when ultimate load tests not practical
- Independent analyses can be used to extend test-based data for similarity verification
- Leverage ALL previous test, analysis, and engineering experience data to the fullest extent to minimize risk
- Designs should 'evolve' to more exotic material and manufacturing

Key Lessons Learned



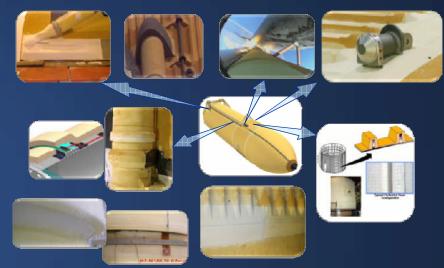


Goal

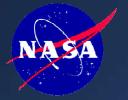
Post-Columbia ET Project directed to *eliminate* critical debris sources by redesign or provide flight rationale

Challenges

- Limited understanding of TPS material, failure modes, and analysis methods
 - TPS debris never considered a 'safety of flight concern' before STS-107



- Extreme amount of distrust / anxiety within technical and programmatic communities
- Schedule pressure to Return to Flight 6 month target for RTF initially established...
- Debris expectations not effectively communicated or understood coupled with unexpected foam loss on RTF I
- Limited ability to effectively communicate integrated debris risk
- TPS performance after RTF I resulted in stand-down to perform additional debris mitigations
- Hurricane Katrina devastated south Louisiana following RTF I
- TPS scope increase from RTF I and RTF II jeopardized ET's ability to 'Meet the Manifest'

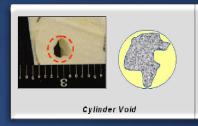


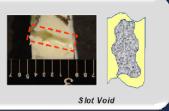


How'd We Do It?

Performed extensive test program to document and characterize TPS application process performance







Debris
Allowable Mass

Possible
Divots

Critical Debris Risk

Allowable Debris

Defect Critical Dimension

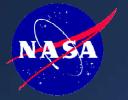
Performed failure mode tests to relate process data to debris expectations

Enhanced flight imagery assets and post flight assessment process to further correlate understanding of failure modes



ET-120 dissection following cryogenic loading provided key insight into TPS failure modes

Significantly improved understanding and communication of debris expectations
- - Fully engaged technical community





How'd We Do It?

Improved understanding of performance led to,

- Improved communication of expectations
- Enhanced designs with improved process controls
- Design / process changes prioritized and implemented as soon as practical – Mod tanks and In-line tanks



Outstanding dedication and perseverance in the face of extreme hardships

- A section of eastern New Orleans after Hurricane Katrina
- Michoud Assembly Facility (green) is not flooded, while the surrounding neighborhoods (dark greenish brown) are extensively flooded







ET Team become extremely resilient and efficient when confronted with difficult situations..



STS-118 8/8



201

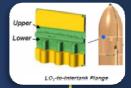
Challenges and learning led to continuous improvements during RTF

ET- 121 RTF Modifications
Bipod Fitting and Feedline Bellows Heater

ET- 124 Hail Damaged Tank ET -129/127/130/131
TPS Producibility Enhancements











ET- 119 RTF II Modifications PAL Ramp Removal



ET-120 LH2 IFR
Redesign Demonstration



STS-123 3/11

ET- 128, First In-line Tank
LO2 Feedline Brackets and LH2 IFR's

Processing Improvements

- Low Output Spray Guns
- Human Factors
- High Fidelity Mockups
- Video Review Of Sprays
- Improved Tank Access
- NDE
- Produciblity Enhancements
- GUCP Improvements
- Friction Stir Welding

Design Improvements

- Bipod Fitting
- Bellows Heater
- Feedline Camera
- PAL Removal
- LH IFR's
- LO2 Ti Brackets
- ECO Feed Through
- Sixth Buy Tanks

Post Flight Assessment

- Imagery
- Failure Mode Assessment
- CAD Modeling
- EPAT Process
- Historical Data Base
- Statistical Assessments







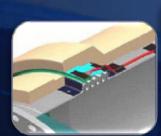
Remove/Replace Longeron Closeouts



Feedline Fairing Camera



Bipod Fitting Closeout



Feedline Bellows



Forward feedline Bellows Heater



Intertank/LH2 Tank Flange Closeout Enhancement



Increase Area of Vented Intertank TPS



Without heater



RTF I: ET-121 modifications to reduce foam and ice debris

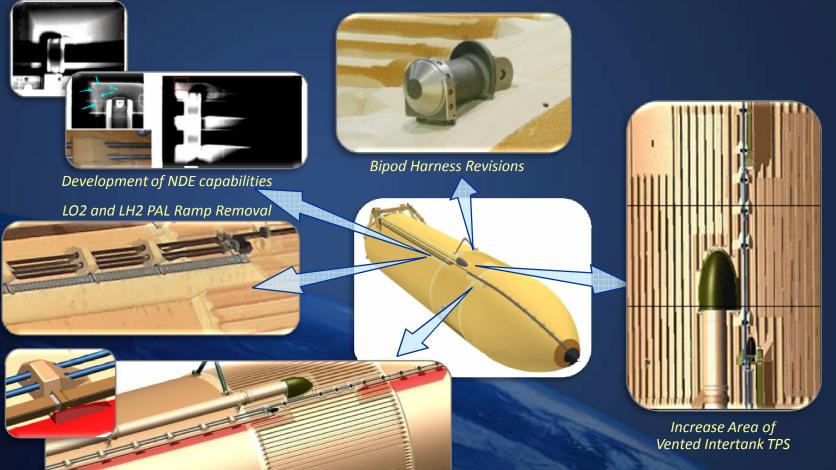
-- Performance improved but 'unexpected' observations from the LH2 flange, bipod, and LH2 PAL ramp resulted in stand-down and RTF II



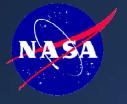
NASA

External Tank Legacy of Success Return to Flight





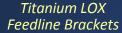
RTF II: Following STS-114, additional foam debris risk mitigation
-- Elimination of the PAL ramps and Bipod heater harness modifications were key changes

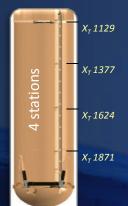




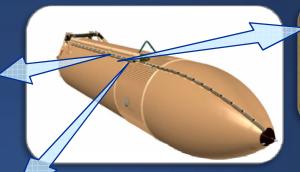
1151

1205











1270 1334 1399 1464 1528 1593 1657 1722 1787 1851 1916 1980 2013/2028

Proposed fine

Zero Gap / Slip Plane (Teflon)



Aluminum Bracket



Titanium Bracket





Cryopumping path along shear pin hole eliminated — Issue identified through ET-120 dissections and testing

Beginning with ET-128 additional 'in-line' changes implemented to further improve debris performance
--LH2 IFR's and LOX feedline titanium brackets





 Post-RTF process and design changes have increased work scope and complexity



& TPS Closeout

LO2 Feedline Support Bracket TPS

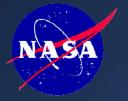


SSP Requirements and ET Planning						
ET	STS	SSP Ship	Current ET DD250	Revised ET DD250	Improvement Challenges	
ET-127	125	07/02/08	07/24/08	07/15/08	-13	
ET-129	126	08/04/08	08/18/08	08/09/08	-5	
ET-130	119	11/15/08	12/05/08	12/22/08	-37	
ET-131	127	03/11/09	03/23/09	04/03/09	-23	
ET-132	128	05/11/09	06/05/09	06/03/09	-23	
ET-133	129	07/14/09	08/24/09	08/25/09	-42	
ET-134	130	09/14/09	11/03/09	10/09/09	-25	
ET-135	131	11/14/09	01/21/10	01/05/10	-52	
ET-136	132	01/20/10	04/05/10	02/25/10	-36	
ET-137	133	03/23/10	06/04/10	04/26/10	-34	
ET-138	LON	07/10/10	08/26/10	07/09/10	1	

	НБРТА –	RTF Requiremen	Production			
Application	Proficiency# of Sprays	Proficiency Time Req	# of Sprays per Part	Sprays per tank RTF	Sprays per tank Pre-RTF	
Longeron	5	28 days ²	3	10	2	
LH2 Flange	4	28 days	56 ³	28	1	
LO2 Flange	2	28 days	9	9	1	
Bipod	2	28 days	4	4	2	

- 1. Requires detailed dissection and data analysis
- 2. Recently changed from 28 to 90 days based on performance
- 3. Recently changed from 56 to 34 days by eliminating the leadout HFPTA's based on performance

117 Additional Sprays
Performed Per Tank Since RTF
- Proficiency Spray
Requirement adds 13 Sprays
(time dependent)





LO-to-Intertank Flange

 ET Project with the help of Shuttle community aggressively pursued changes to reduce scope without increasing debris potential

Leveraged improved understanding of physics and debris risk

Redesigned TPS
Applications
27 days

Simplified TPS
Trim Processes
5.5 days

KSC Deferred Work 8 days

Expanded Acceptance Criteria 1.5 days

Streamlined Process Controls 1.5 days

Revised Inspection Requirements 4 days

> SLA Deletions 1 day



On-time Deliveries and Outstanding Performance

Process and Design Changes

LO₂-to-Intertank Flange Closeout

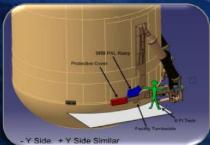


Single Pass LO2/IT Flange Spray Reduced cycle-time by ~25 days in Cell G/H

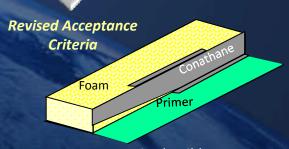
Upper

Lower

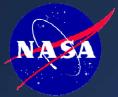




Training / Certification for Deferred Work to KSC



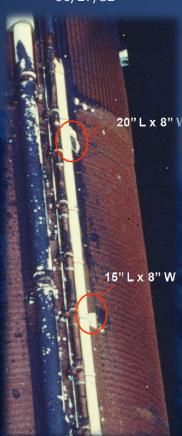
Revised Build Acceptance Criteria has Lowered NCDs



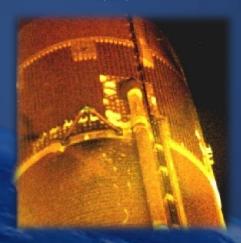


Post RTF TPS Performance Dramatically Better than Previous Designs





STS-26/ET-19 07/29/85



 Intertank foam loss (two tone acreage foam) 18 divots,
 3" to 30" diameter

STS-126 / ET-129 11/14/08



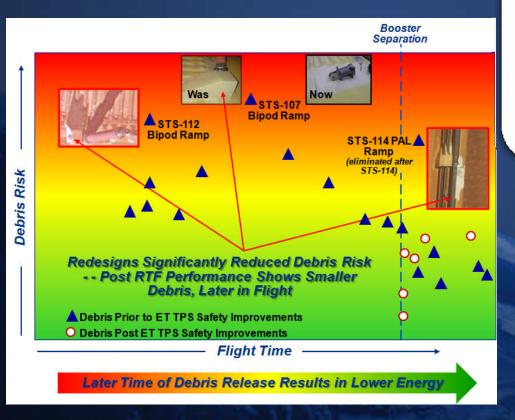


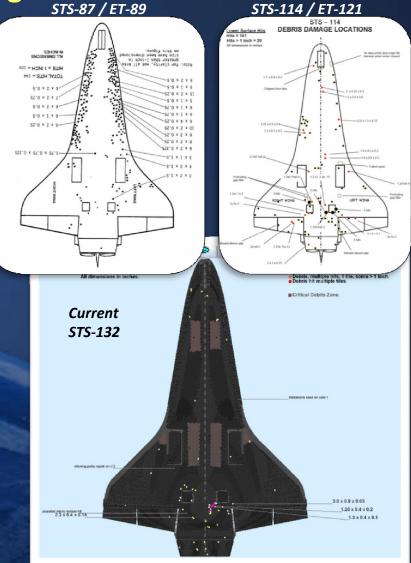
External Tank Legacy of Success

Return to Flight

Post RTF TPS Performance Demonstrates Reduction in Debris Risk

 Recent Orbiter tile damage maps show fewest and smallest impacts over life of program





RTF I

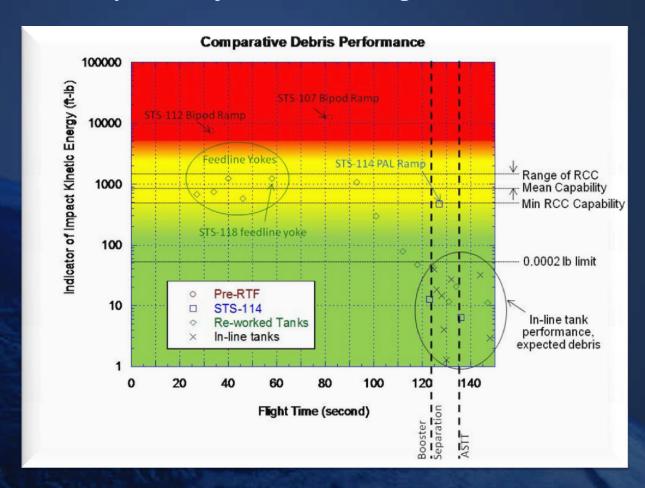


Post RTF Debris Performance Results in Smaller, Later Debris and less Orbiter lower surface damage





Comparison of all Return to Flight Missions



Impact energy continuously reduced as design improvements were implemented







Outstanding External Tank Performance for STS-132
-- Orbiter lower surface damage was the lowest since Return to Flight





- Ensure early and clear understanding of requirements
- Acknowledge verification limitations and effectively communicate risk to program management

Key Lessons Learned

- Ensure a strong physics-based understanding of failure modes with test-demonstration
- Be cautious of implementing excessive process controls without having a good understanding of failure modes
- Critical to communicate and educate all stakeholders early and often
 - Can be difficult but the payoff is HUGE!!

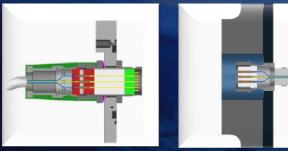


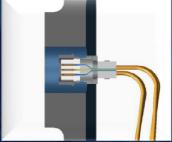
External Tank Legacy of Success ET-125 ECO Sensor Anomaly



- Establish root cause and redesign solution to mitigate LH2 tank engine cut-off (ECO) sensor circuit anomalies
 - Anomalies resulted in multiple launch scrubs since RTF







Feed Through Connector

Depletion Sensors Installed on Shock Mount

Challenges

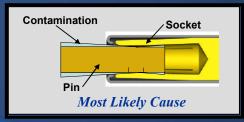
- Intermittent nature of failures and system complexity made troubleshooting difficult
- Data set not compelling enough to focus community many different opinions
- Emerging data issues (feed thru glass cracks / pin contamination)
- Launch pressure to support manifest / ISS construction



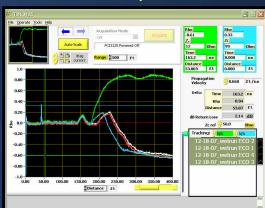
External Tank Legacy of Success ET-125 ECO Sensor Anomaly



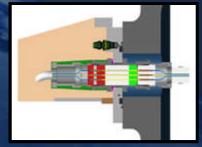
- How'd We Do It?
 - Got Lucky?? Anomaly signature on ET-125 focused efforts on feed thru connector
 - Simultaneous failure of two sensor circuits and timing related to fill level
 - Strong Leadership Decision made by Wayne Hale to stand-down and fix problem
 - Developed physics-based scenario to explain failure signatures



- Outstanding teamwork to perform tanking test with instrumentation to confirm theory and isolate fault
- Leveraged previous experiences from Atlas / Centaur to develop rapid corrective action design solution
- Developed innovative TPS analysis and repairs to minimize ablator applications



Time Domain Reflectometry used to detect fault during tanking test



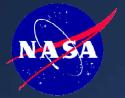
Outstanding teamwork to perform expeditious certification test program of redesign



External Tank Legacy of Success ET-125 ECO Sensor Anomaly



- Results
 - Design change successfully implemented and flown with no issues
 - Rock solid sensor circuit performance
- Key Lessons Learned
 - Physics should drive the investigation and corrective action
 - Originally focused on sensor as source of fault
 - Physics and testing did not support scenario
 - Critical to learn from and incorporate lessons learned
 - Atlas / Centaur corrective actions not fully understood or investigated during initial phase of investigation



External Tank Legacy of Success ET-124 Hail Damage Recovery



Goal

- Extensive TPS damage caused by extreme hail storm
- Repair plan required to restore TPS to minimize program manifest impacts

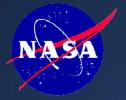
Challenges

- Skeptical technical community Concerned about interactions of damage with known / unknown failure modes
- Schedule pressure to accommodate ISS program
 - Next tank still at MAF
- Limited ET resources

How'd We Do It?

- Developed unique engineering requirements and tooling to minimize repairs
- Performed large amount of performance testing to demonstrate understanding of repairs and residual conditions
- Effectively communicated results to technical community and management to instill confidence in expected performance



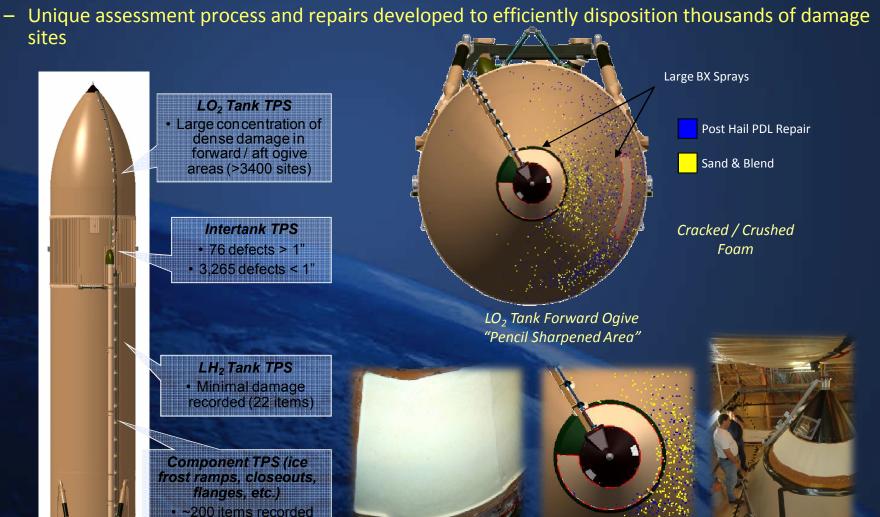


External Tank Legacy of Success ET-124 Hail Damage Recovery

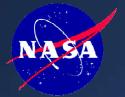


How'd We Do It?





Primarily superficial



External Tank Legacy of Success ET-124 Hail Damage Recovery



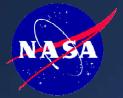
Results

- STS-117/ET-124 scheduled to Launch 03/15/07
 - Repaired and successfully launched 6/8/07
- Post flight Assessment revealed no issues with any of the hail damage repairs
- Success enables ET Project to improve 'credibility' within technical community
 - Very helpful with future changes to enhance producibility

Key Lessons Learned

- Be creative in the face of adversity
 - Explore requirements to capitalize on any mission unique aspects
- Tests are critical to demonstrate performance and minimize anxiety / chaos
- Communicate early and often





External Tank Legacy of Success STS-130 / ET-134 Launch



- Goal
 - Launch STS-130 / ET-134 2/8/10

 2/8 launch date required due to 'suspicious' launch delay from the previous day due to low 'Colts blue' debris clouds

Challenges

- Saints in Super Bowl!!!!
- Call to stations / launch support coincided with kick-off
- Large group of 'Who Dats' required for launch support
- Pressure to keep up with BCS champs Alabama!

How'd We Do It?

 We did it – A day the ET Team will never forget..

Keep Your Teams Focused and FINISH STRONG



